

# Rayleigh variational principle and vibrations of prestressed shells

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# Outline

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# General theory of shells

Here we use the general theory of shells presented e.g. in<sup>1,2,3</sup>

- The kinematics of the shell is determined by two kinematically independent fields of translations and rotations. 6 degrees of freedom: each point of the micropolar shell base surface has six degrees of freedom as in the rigid body dynamics.
- At the shell boundary acts forces and moments only.
- The drilling moment is also taken in account.
- Using certain constraints one can reduce the micropolar shell theory to the Kirchhoff–Love or Reissner–Mindlin shell models.

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<sup>1</sup>A. Libai, J. G. Simmonds, *The Nonlinear Theory of Elastic Shells*, 2nd Edition, Cambridge University Press, Cambridge, 1998.

<sup>2</sup>J. Chróścielewski, J. Makowski, W. Pietraszkiewicz, *Statics and Dynamics of Multifolded Shells. Nonlinear Theory and Finite Element Method* (in Polish), Wydawnictwo IPPT PAN, Warszawa, 2004.

<sup>3</sup>V. A. Eremeyev, L. P. Lebedev, H. Altenbach, *Foundations of Micropolar Mechanics*, Springer, Heidelberg, 2012.

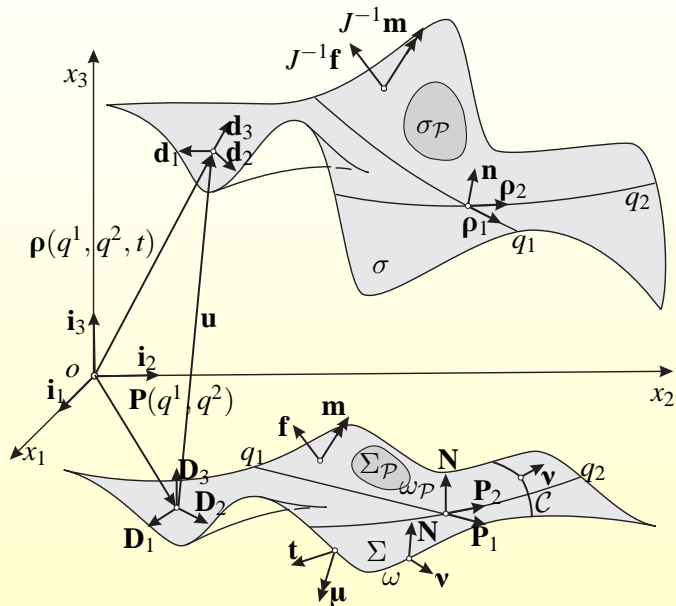
# Shell Kinematics

- The deformation of the shell is described by mapping from one state called the *reference configuration* to another one called the *actual configuration*.
- In the reference configuration  $\varkappa: \Sigma$  is a base surface with position vector  $\mathbf{P}(q^1, q^2)$  and with directors  $\mathbf{D}_k(q^1, q^2, \cdot)$ ,  $k = 1, 2, 3$ .
- In the actual configuration  $\chi: \sigma$  is the base surface with position vector  $\boldsymbol{\rho}(q^1, q^2, t)$  and directors  $\mathbf{d}_k(q^1, q^2, t)$ ,  $k = 1, 2, 3$ .

Hence, the shell is described by two kinematically independent fields

$$\boldsymbol{\rho} = \boldsymbol{\rho}(q^1, q^2, t) \quad \text{and} \quad \mathbf{Q} \equiv \mathbf{d}_k \otimes \mathbf{D}_k = \mathbf{Q}(q^1, q^2, t). \quad (1)$$

# Shell Kinematics



# Constitutive Equations of Elastic Shells (1)

According to the local action principle <sup>4</sup>, the strain energy density  $W$  takes the form

$$W = W(\boldsymbol{\rho}, \nabla_{\mathcal{X}}\boldsymbol{\rho}, \mathbf{Q}, \nabla_{\mathcal{X}}\mathbf{Q}),$$

where

$$\nabla_{\mathcal{X}} \triangleq \mathbf{P}^{\alpha} \frac{\partial}{\partial q^{\alpha}} \quad (\alpha, \beta = 1, 2),$$

$$\mathbf{P}^{\alpha} \cdot \mathbf{P}_{\beta} = \delta_{\beta}^{\alpha}, \quad \mathbf{P}^{\alpha} \cdot \mathbf{N} = 0, \quad \mathbf{P}_{\beta} = \frac{\partial \mathbf{P}}{\partial q^{\beta}}.$$

Here vectors  $\mathbf{P}_{\beta}$  and  $\mathbf{P}^{\alpha}$  denote the natural and reciprocal bases on  $\Sigma$ , respectively,  $\mathbf{N}$  is the unit normal to  $\Sigma$ ,  $\delta_{\beta}^{\alpha}$  is the Kronecker symbol, and  $\nabla_{\mathcal{X}}$  is the surface nabla operator on  $\Sigma$ .

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<sup>4</sup>C. Truesdell, W. Noll, The nonlinear field theories of mechanics, in: S. Flügge (Ed.), Handbuch der Physik, Vol. III/3, Springer, Berlin, 1965, pp. 1–602.

## Constitutive Equations of Elastic Shells (2)

From the principle of material frame-indifference <sup>5</sup> it follows that  $W$  depends on two surface strain measures  $\mathbf{E}$  and  $\mathbf{K}$ :

$$W = W(\mathbf{E}, \mathbf{K}), \quad (2)$$

where

$$\mathbf{E} = \mathbf{F} \cdot \mathbf{Q}^T - \mathbf{A}, \quad \mathbf{K} = \frac{1}{2} \mathbf{P}^\alpha \otimes \left( \frac{\partial \mathbf{Q}}{\partial q^\alpha} \cdot \mathbf{Q}^T \right)_\times. \quad (3)$$

Here  $\mathbf{F} = \nabla_{\mathcal{S}} \boldsymbol{\rho}$ , and  $\mathbf{A} \triangleq \mathbf{I} - \mathbf{N} \otimes \mathbf{N}$ ,  $\mathbf{I}$  is the unit 3D tensor,

$$\mathbf{T}_\times = (T^{mn} \mathbf{i}_m \otimes \mathbf{i}_n)_\times = T^{mn} \mathbf{i}_m \times \mathbf{i}_n$$

for any base  $\mathbf{i}_m$ ,  $\times$  denotes the vector (cross) product.

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<sup>5</sup>C. Truesdell, W. Noll, The nonlinear field theories of mechanics, in: S. Flügge (Ed.), Handbuch der Physik, Vol. III/3, Springer, Berlin, 1965, pp. 1–602.

# Vectorial Parameterizations of Strain Measures

Introducing

- the translation vector  $\mathbf{u} = \boldsymbol{\rho} - \mathbf{P}$  and
- the finite rotation vector  $\boldsymbol{\theta} = 2\mathbf{e} \tan \varphi/2$

we can express  $\mathbf{Q}$ ,  $\mathbf{E}$  and  $\mathbf{K}$  as follows (see <sup>6</sup> for details)

$$\mathbf{Q} = \frac{1}{(4 + \theta^2)} [(4 - \theta^2)\mathbf{I} + 2\boldsymbol{\theta} \otimes \boldsymbol{\theta} - 4\mathbf{I} \times \boldsymbol{\theta}], \quad (4)$$

$$\mathbf{E} = (\mathbf{A} + \nabla_{\boldsymbol{x}}\mathbf{u}) \cdot \mathbf{Q}^T - \mathbf{A}, \quad \theta^2 = \boldsymbol{\theta} \cdot \boldsymbol{\theta}, \quad (5)$$

$$\mathbf{K} = \frac{4}{4 + \theta^2} \nabla_{\boldsymbol{x}}\boldsymbol{\theta} \cdot \left( \mathbf{I} + \frac{1}{2}\mathbf{I} \times \boldsymbol{\theta} \right). \quad (6)$$

$\mathbf{Q}$  describes the rotation about axis with the unit vector  $\mathbf{e}$  through an angle  $\varphi$ .

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<sup>6</sup>W. Pietraszkiewicz, V. A. Eremeyev, On vectorially parameterized natural strain measures of the non-linear Cosserat continuum, International Journal of Solids and Structures 46 (11–12) (2009) 2477–2480.

# Lagrangian Equations of Motion

The Lagrangian equations of motion of the micropolar shell are

$$\nabla_{\mathcal{X}} \cdot \mathbf{T}_{\mathcal{X}} + \mathbf{f} = \rho_{\mathcal{X}} \frac{d\mathbf{K}_1}{dt}, \quad (7)$$

$$\nabla_{\mathcal{X}} \cdot \mathbf{M}_{\mathcal{X}} + [\mathbf{F}^T \cdot \mathbf{T}_{\mathcal{X}}]_{\times} + \mathbf{m} = \rho_{\mathcal{X}} \left( \frac{d\mathbf{K}_2}{dt} + \mathbf{v} \times \boldsymbol{\Theta}_1^T \cdot \boldsymbol{\omega} \right), \quad (8)$$

where

$$\mathbf{T}_{\mathcal{X}} = \mathbf{S}_1 \cdot \mathbf{Q}, \quad \mathbf{M}_{\mathcal{X}} = \mathbf{S}_2 \cdot \mathbf{Q}, \quad (9)$$

$$\mathbf{S}_1 = \frac{\partial W}{\partial \mathbf{E}}, \quad \mathbf{S}_2 = \frac{\partial W}{\partial \mathbf{K}}, \quad (10)$$

$$\mathbf{K}_1 = \frac{\partial K}{\partial \mathbf{v}} = \mathbf{v} + \boldsymbol{\Theta}_1^T \cdot \boldsymbol{\omega}, \quad \mathbf{K}_2 = \frac{\partial K}{\partial \boldsymbol{\omega}} = \boldsymbol{\Theta}_1 \cdot \mathbf{v} + \boldsymbol{\Theta}_2 \cdot \boldsymbol{\omega}, \quad (11)$$

$$K(\mathbf{v}, \boldsymbol{\omega}) = \frac{1}{2} \mathbf{v} \cdot \mathbf{v} + \boldsymbol{\omega} \cdot \boldsymbol{\Theta}_1 \cdot \mathbf{v} + \frac{1}{2} \boldsymbol{\omega} \cdot \boldsymbol{\Theta}_2 \cdot \boldsymbol{\omega}, \quad (12)$$

$$\mathbf{v} = \frac{d\boldsymbol{\rho}}{dt}, \quad \boldsymbol{\omega} = \frac{1}{2} \left( \mathbf{Q}^T \cdot \frac{d\mathbf{Q}}{dt} \right)_{\times}.$$

# Eulerian Equations of Motion

The Eulerian equations of motion of the micropolar shell are

$$\nabla_{\chi} \cdot \mathbf{T} + J^{-1} \mathbf{f} = \rho \frac{d\mathbf{K}_1}{dt}, \quad (13)$$

$$\nabla_{\chi} \cdot \mathbf{M} + \mathbf{T}_{\times} + J^{-1} \mathbf{m} = \rho \left( \frac{d\mathbf{K}_2}{dt} + \mathbf{v} \times \boldsymbol{\Theta}_1^T \cdot \boldsymbol{\omega} \right), \quad (14)$$

where

$$\nabla_{\chi} = \boldsymbol{\rho}^{\alpha} \frac{\partial}{\partial q^{\alpha}}, \quad \boldsymbol{\rho}^{\alpha} \cdot \boldsymbol{\rho}^{\beta} = \delta_{\beta}^{\alpha}, \quad \boldsymbol{\rho}^{\alpha} \cdot \mathbf{n} = 0, \quad \boldsymbol{\rho}^{\beta} = \frac{\partial \boldsymbol{\rho}}{\partial q^{\beta}},$$
$$\mathbf{T} = J^{-1} \mathbf{F}^T \cdot \mathbf{T}_{\varkappa}, \quad \mathbf{M} = J^{-1} \mathbf{F}^T \cdot \mathbf{M}_{\varkappa}, \quad (15)$$

$$J = J(\mathbf{F}) = \sqrt{\frac{1}{2} \left\{ [\text{tr}(\mathbf{F} \cdot \mathbf{F}^T)]^2 - \text{tr}[(\mathbf{F} \cdot \mathbf{F}^T)^2] \right\}}.$$

Here  $\mathbf{T}$  and  $\mathbf{M}$  are Cauchy-type surface stress and couple stress tensors,  $\rho$  is the surface mass density in the actual configuration,  $\nabla_{\chi}$  is the surface nabla operator on  $\sigma$  related with  $\nabla_{\varkappa}$  by the formula  $\nabla_{\varkappa} = \mathbf{F} \cdot \nabla_{\chi}$ , and  $\mathbf{n}$  is the unit normal to  $\sigma$ .

# Initial Boundary-Value Problem

Simplifications

$$\Theta_1 = \mathbf{0}, \quad \Theta_2 = \gamma \mathbf{I}, \quad (16)$$

where  $\gamma$  is a scalar measure of the rotatory inertia. The equations of motion (7) and (8) take more simple form

$$\nabla_{\mathcal{X}} \cdot \mathbf{T}_{\mathcal{X}} + \mathbf{f} = \rho_{\mathcal{X}} \frac{d\mathbf{v}}{dt}, \quad (17)$$

$$\nabla_{\mathcal{X}} \cdot \mathbf{M}_{\mathcal{X}} + [\mathbf{F}^T \cdot \mathbf{T}_{\mathcal{X}}]_{\mathcal{X}} + \mathbf{m} = \rho_{\mathcal{X}} \gamma \frac{d\boldsymbol{\omega}}{dt}. \quad (18)$$

Equations of motion are supplemented by the boundary conditions

$$\begin{aligned} \text{on } \omega_1 : \quad \boldsymbol{\rho} &= \mathbf{r}_0(s), & \text{on } \omega_2 : \quad \mathbf{v} \cdot \mathbf{T}_{\mathcal{X}} &= \mathbf{t}(s), \\ \text{on } \omega_3 : \quad \mathbf{Q} &= \mathbf{h}(s), & \text{on } \omega_4 : \quad \mathbf{v} \cdot \mathbf{M}_{\mathcal{X}} &= \boldsymbol{\mu}(s), \end{aligned} \quad (19)$$

and initial conditions

$$\boldsymbol{\rho}|_{t=0} = \boldsymbol{\rho}^\circ, \quad \mathbf{v}|_{t=0} = \mathbf{v}^\circ, \quad \mathbf{Q}|_{t=0} = \mathbf{Q}^\circ, \quad \boldsymbol{\omega}|_{t=0} = \boldsymbol{\omega}^\circ. \quad (20)$$

with given initial values  $\boldsymbol{\rho}^\circ$ ,  $\mathbf{v}^\circ$ ,  $\mathbf{Q}^\circ$ ,  $\boldsymbol{\omega}^\circ$ , and given functions  $\mathbf{r}_0(s)$ ,  $\mathbf{h}(s)$ ,  $\mathbf{t}(s)$ , and  $\boldsymbol{\mu}(s)$ .

# Isotropic Shell

Physically linear isotropic shell

$$\begin{aligned} 2W &= \alpha_1 \text{tr}^2 \mathbf{E}_{\parallel} + \alpha_2 \text{tr} \mathbf{E}_{\parallel}^2 + \alpha_3 \text{tr} \left( \mathbf{E}_{\parallel} \cdot \mathbf{E}_{\parallel}^T \right) + \alpha_4 \mathbf{N} \cdot \mathbf{E}^T \cdot \mathbf{E} \cdot \mathbf{N} \\ &+ \beta_1 \text{tr}^2 \mathbf{K}_{\parallel} + \beta_2 \text{tr} \mathbf{K}_{\parallel}^2 + \beta_3 \text{tr} \left( \mathbf{K}_{\parallel} \cdot \mathbf{K}_{\parallel}^T \right) + \beta_4 \mathbf{N} \cdot \mathbf{K}^T \cdot \mathbf{K} \cdot \mathbf{N}, \end{aligned} \quad (21)$$

where  $\mathbf{E}_{\parallel} \triangleq \mathbf{E} \cdot \mathbf{A}$ ,  $\mathbf{K}_{\parallel} \triangleq \mathbf{K} \cdot \mathbf{A}$ . Here  $\mathbf{S}_1$  and  $\mathbf{S}_2$  have the form

$$\mathbf{S}_1 = \alpha_1 (\text{tr} \mathbf{E}_{\parallel}) \mathbf{A} + \alpha_2 \mathbf{E}_{\parallel}^T + \alpha_3 \mathbf{E}_{\parallel} + \alpha_4 (\mathbf{E} \cdot \mathbf{N}) \otimes \mathbf{N}, \quad (22)$$

$$\mathbf{S}_2 = \beta_1 (\text{tr} \mathbf{K}_{\parallel}) \mathbf{A} + \beta_2 \mathbf{K}_{\parallel}^T + \beta_3 \mathbf{K}_{\parallel} + \beta_4 (\mathbf{K} \cdot \mathbf{N}) \otimes \mathbf{N}. \quad (23)$$

Introducing the fourth-order tensors  $\mathbf{C}_1$  and  $\mathbf{C}_2$  by the formulae

$$\begin{aligned} \mathbf{C}_1 &= \alpha_1 \mathbf{A} \otimes \mathbf{A} + \alpha_2 \mathbf{P}_{\alpha} \otimes \mathbf{A} \otimes \mathbf{P}^{\alpha} \\ &+ \alpha_3 \mathbf{P}_{\alpha} \otimes \mathbf{P}_{\beta} \otimes \mathbf{P}^{\alpha} \otimes \mathbf{P}^{\beta} + \alpha_4 \mathbf{P}_{\alpha} \otimes \mathbf{N} \otimes \mathbf{P}^{\alpha} \otimes \mathbf{N}, \end{aligned}$$

$$\begin{aligned} \mathbf{C}_2 &= \beta_1 \mathbf{A} \otimes \mathbf{A} + \beta_2 \mathbf{P}_{\alpha} \otimes \mathbf{A} \otimes \mathbf{P}^{\alpha} \\ &+ \beta_3 \mathbf{P}_{\alpha} \otimes \mathbf{P}_{\beta} \otimes \mathbf{P}^{\alpha} \otimes \mathbf{P}^{\beta} + \beta_4 \mathbf{P}_{\alpha} \otimes \mathbf{N} \otimes \mathbf{P}^{\alpha} \otimes \mathbf{N}, \end{aligned}$$

we re-write (22) and (23) in a compact form  $\mathbf{S}_1 = \mathbf{C}_1 : \mathbf{E}$ ,  $\mathbf{S}_2 = \mathbf{C}_2 : \mathbf{K}$ .

# Superimposed Infinitesimal Deformations

Let  $\rho_0$  and  $\mathbf{Q}_0$  are the known static solution.  
Superimposed infinitesimal deformations are

$$\rho_* = \rho_0 + \delta\rho, \quad \mathbf{Q}_* = \mathbf{Q}_0 + \delta\mathbf{Q},$$

Since  $\mathbf{Q}$  is an orthogonal tensor, the tensor  $\mathbf{Q}^T \cdot \delta\mathbf{Q}$  is

$$\mathbf{Q}^T \cdot \delta\mathbf{Q} = -\mathbf{I} \times \boldsymbol{\psi},$$

where  $\boldsymbol{\psi}$  is the infinitesimal rotation vector expressed by

$$\boldsymbol{\psi} = \frac{4}{4 + \theta^2} \left( \delta\boldsymbol{\theta} + \frac{1}{2}\boldsymbol{\theta} \times \delta\boldsymbol{\theta} \right).$$

The increments of the strain measures are given by the formulae

$$\delta\mathbf{E} = (\nabla_{\boldsymbol{x}} \delta\rho) \cdot \mathbf{Q}_0^T + \mathbf{F}_0 \cdot \delta\mathbf{Q}^T = \mathbf{F}_0 \cdot \boldsymbol{\varepsilon} \cdot \mathbf{Q}_0^T, \quad (24)$$

$$\delta\mathbf{K} = (\nabla_{\boldsymbol{x}} \boldsymbol{\psi}) \cdot \mathbf{Q}^T = \mathbf{F}_0 \cdot \boldsymbol{\varkappa} \cdot \mathbf{Q}_0^T, \quad (25)$$

where  $\boldsymbol{\varepsilon}$  and  $\boldsymbol{\varkappa}$  are the linear strain measures given by

$$\boldsymbol{\varepsilon} = \nabla_{\boldsymbol{\chi}} \mathbf{w} + \mathbf{A} \times \boldsymbol{\psi}, \quad \boldsymbol{\varkappa} = \nabla_{\boldsymbol{\chi}} \boldsymbol{\psi}, \quad (26)$$

# Linearization

## Lagrangian linearized equations of motion

$$\nabla_{\mathcal{X}} \cdot \delta \mathbf{T}_{\mathcal{X}} = \rho_{\mathcal{X}} \frac{d^2 \mathbf{w}}{dt^2}, \quad (27)$$

$$\nabla_{\mathcal{X}} \cdot \delta \mathbf{M}_{\mathcal{X}} + [(\nabla_{\mathcal{X}} \mathbf{w})^T \cdot \mathbf{T}_{\mathcal{X}} + \mathbf{F}_0^T \cdot \delta \mathbf{T}_{\mathcal{X}}]_{\mathcal{X}} = \rho_{\mathcal{X}} \gamma \frac{d^2 \psi}{dt^2}, \quad (28)$$

## Linearized boundary conditions

$$\begin{array}{ll} \text{on } \omega_1 : & \mathbf{w} = \mathbf{0}, \\ \text{on } \omega_2 : & \mathbf{v} \cdot \delta \mathbf{T}_{\mathcal{X}} = \mathbf{0}, \\ \text{on } \omega_3 : & \psi = \mathbf{0}, \\ \text{on } \omega_4 : & \mathbf{v} \cdot \delta \mathbf{M}_{\mathcal{X}} = \mathbf{0}. \end{array} \quad (29)$$

# Linearized Constitutive Relations

$$\delta \mathbf{T}_\varkappa = \delta \mathbf{S}_1 \cdot \mathbf{Q}_0 + \mathbf{S}_1 \cdot \delta \mathbf{Q} = \delta \mathbf{S}_1 \cdot \mathbf{Q}_0 - \mathbf{T}_\varkappa \times \boldsymbol{\psi}, \quad (30)$$

$$\delta \mathbf{M}_\varkappa = \delta \mathbf{S}_2 \cdot \mathbf{Q}_0 + \mathbf{S}_2 \cdot \delta \mathbf{Q} = \delta \mathbf{S}_2 \cdot \mathbf{Q}_0 - \mathbf{M}_\varkappa \times \boldsymbol{\psi}, \quad (31)$$

$$\delta \mathbf{S}_1 = \frac{\partial W}{\partial \mathbf{E} \partial \mathbf{E}} : \delta \mathbf{E} + \frac{\partial W}{\partial \mathbf{E} \partial \mathbf{K}} : \delta \mathbf{K}, \quad (32)$$

$$\delta \mathbf{S}_2 = \frac{\partial W}{\partial \mathbf{K} \partial \mathbf{E}} : \delta \mathbf{E} + \frac{\partial W}{\partial \mathbf{K} \partial \mathbf{K}} : \delta \mathbf{K}. \quad (33)$$

For the physically linear shell we have

$$\delta \mathbf{S}_1 = \mathbf{C}_1 : \delta \mathbf{E} = \mathbf{D}_1 : \boldsymbol{\varepsilon}, \quad \delta \mathbf{S}_2 = \mathbf{C}_2 : \delta \mathbf{K} = \mathbf{D}_2 : \boldsymbol{\varkappa},$$

where  $\mathbf{D}_1$  and  $\mathbf{D}_2$  are the fourth-order tensors given by

$$\begin{aligned} \mathbf{D}_1 &= \alpha_1 \mathbf{A} \otimes \mathbf{F}_0^T \cdot \mathbf{P}_\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\alpha + \alpha_2 \mathbf{P}_\alpha \otimes \mathbf{P}_\beta \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\beta \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\alpha \\ &\quad + \alpha_3 \mathbf{P}_\alpha \otimes \mathbf{P}_\beta \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\beta + \alpha_4 \mathbf{P}_\alpha \otimes \mathbf{N} \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{N}, \\ \mathbf{D}_2 &= \beta_1 \mathbf{A} \otimes \mathbf{F}_0^T \cdot \mathbf{P}_\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\alpha + \beta_2 \mathbf{P}_\alpha \otimes \mathbf{P}_\beta \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\beta \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\alpha \\ &\quad + \beta_3 \mathbf{P}_\alpha \otimes \mathbf{P}_\beta \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{P}^\beta + \beta_4 \mathbf{P}_\alpha \otimes \mathbf{N} \otimes \mathbf{F}_0^T \cdot \mathbf{P}^\alpha \otimes \mathbf{Q}_0^T \cdot \mathbf{N}. \end{aligned}$$

# Linearized Eulerian Equations of Motion and BCs

Introducing the tensors

$$\Phi_1 = J_0^{-1} \mathbf{F}_0^T \cdot \delta \mathbf{T}_\varkappa, \quad \Phi_2 = J_0^{-1} \mathbf{F}_0^T \cdot \delta \mathbf{M}_\varkappa, \quad (34)$$

where  $J_0 = J(\mathbf{F}_0)$ , we transform (27) and (28) into the linearized equation of motion in the actual configuration  $\chi_0$

$$\nabla_\chi \cdot \Phi_1 = \rho \frac{d^2 \mathbf{w}}{dt^2}, \quad (35)$$

$$\nabla_\chi \cdot \Phi_2 + [(\nabla_\chi \mathbf{w})^T \cdot \mathbf{T} + \Phi_1]_\times = \rho \gamma \frac{d^2 \psi}{dt^2}. \quad (36)$$

For the physically linear isotropic shell  $\Phi_1$  and  $\Phi_2$  are

$$\begin{aligned} \Phi_1 &= \mathbf{H}_1 : \varepsilon - \mathbf{T} \times \psi, & \Phi_2 &= \mathbf{H}_2 : \varkappa - \mathbf{M} \times \psi, \\ \mathbf{H}_1 &= J_0^{-1} \mathbf{F}_0^T \cdot \mathbf{D}_1, & \mathbf{H}_2 &= J_0^{-1} \mathbf{F}_0^T \cdot \mathbf{D}_2. \end{aligned}$$

The linearized Eulerian boundary conditions are

$$\begin{aligned} \text{on } \ell_1 : \quad \mathbf{w} &= \mathbf{0}, & \text{on } \ell_2 : \quad \boldsymbol{\eta} \cdot \Phi_1 &= \mathbf{0}, \\ \text{on } \ell_3 : \quad \psi &= \mathbf{0}, & \text{on } \ell_4 : \quad \boldsymbol{\eta} \cdot \Phi_2 &= \mathbf{0}. \end{aligned} \quad (37)$$

# Free Vibration of Prestresses Shell

$$\mathbf{w} = \mathbf{W}(q^1, q^2)e^{i\Omega t}, \quad \psi = \Psi(q^1, q^2)e^{i\Omega t}.$$

Substituting the latter relations into (35) and (37) we obtain the boundary-value problem for the physically linear isotropic prestressed micropolar shell

$$\nabla_{\chi} \cdot \Phi_1 = -\rho\Omega^2 \mathbf{W}, \quad (38)$$

$$\nabla_{\chi} \cdot \Phi_2 + [(\nabla_{\chi} \mathbf{w})^T \cdot \mathbf{T} + \Phi_1]_{\chi} = -\rho\gamma\Omega^2 \Psi, \quad (39)$$

$$\begin{aligned} \text{on } l_1 : \mathbf{W} &= \mathbf{0}, & \text{on } l_2 : \boldsymbol{\eta} \cdot \Phi_1 &= \mathbf{0}, \\ \text{on } l_3 : \Psi &= \mathbf{0}, & \text{on } l_4 : \boldsymbol{\eta} \cdot \Phi_2 &= \mathbf{0}, \end{aligned} \quad (40)$$

where

$$\begin{aligned} \Phi_1 &= \mathbf{H}_1 : \boldsymbol{\varepsilon} - \mathbf{T} \times \Psi, & \Phi_2 &= \mathbf{H}_2 : \boldsymbol{\varkappa} - \mathbf{M} \times \Psi, \\ \boldsymbol{\varepsilon} &= \nabla_{\chi} \mathbf{W} + \mathbf{A} \times \Psi, & \boldsymbol{\varkappa} &= \nabla_{\chi} \Psi. \end{aligned} \quad (41)$$

# Comparison Problem

Additionally we consider the linear boundary-value problem of the micropolar shell without initial deformation, that is when  $\chi_0 = \varkappa$ , which is given by

$$\nabla_{\chi} \cdot \Phi_1^0 = -\rho\Omega^2 \mathbf{W}, \quad \nabla_{\chi} \cdot \Phi_2^0 + \Phi_{1 \times}^0 = -\rho\gamma\Omega^2 \Psi, \quad (42)$$

$$\begin{aligned} \text{on } \ell_1 : \quad \mathbf{W} &= \mathbf{0}, & \text{on } \ell_2 : \quad \boldsymbol{\eta} \cdot \Phi_1^0 &= \mathbf{0}, \\ \text{on } \ell_3 : \quad \Psi &= \mathbf{0}, & \text{on } \ell_4 : \quad \boldsymbol{\eta} \cdot \Phi_2^0 &= \mathbf{0}, \end{aligned} \quad (43)$$

$$\Phi_1^0 = \mathbf{C}_1 : \boldsymbol{\varepsilon}, \quad \Phi_1^0 = \mathbf{C}_2 : \varkappa. \quad (44)$$

Comparison of  $\Phi_1^0$  and  $\Phi_1$ ,  $\Phi_2^0$  and  $\Phi_2$  shows that difference between these boundary-value problems consists of

- 1 difference between the elastic moduli tensors  $\mathbf{C}_\alpha$  and  $\mathbf{H}_\alpha$ ,  $\alpha = 1, 2$ , and
- 2 existence of initial stress tensors  $\mathbf{T}$  and  $\mathbf{M}$  in  $\Phi_1$  and  $\Phi_2$ .

In what follows we show the influence on the eigen-frequencies of the prestressed shell using the variational approach.

## Second Variation of the Total Energy

The total potential energy of the shell is

$$\Pi = \iint_{\Sigma} W d\Sigma - \iint_{\Sigma} \mathbf{f} \cdot \mathbf{u} d\Sigma - \int_{\omega_2} \mathbf{t} \cdot \mathbf{u} ds.$$

The second variation of energy is

$$\delta^2 \Pi = 2 \iint_{\sigma} w d\sigma, \quad w = w_1 + w_2, \quad (45)$$

where

$$\begin{aligned} w_1(\boldsymbol{\varepsilon}, \boldsymbol{\varkappa}) &= \frac{1}{2} \boldsymbol{\varepsilon} : \mathbf{H}_1 : \boldsymbol{\varepsilon} + \frac{1}{2} \boldsymbol{\varkappa} : \mathbf{H}_2 : \boldsymbol{\varkappa}, \\ w_2(\boldsymbol{\psi}, \boldsymbol{\varepsilon}, \boldsymbol{\varkappa}) &= \operatorname{tr} (\boldsymbol{\psi} \times \mathbf{T}^T \cdot \boldsymbol{\varepsilon}) - \frac{1}{2} \operatorname{tr} (\boldsymbol{\psi} \times \mathbf{T}^T \times \boldsymbol{\psi}) + \frac{1}{2} \operatorname{tr} (\boldsymbol{\psi} \times \mathbf{M}^T \cdot \boldsymbol{\varkappa}). \end{aligned} \quad (46)$$

If  $\chi_0 = \boldsymbol{\varkappa}$ , that is  $\mathbf{T} = \mathbf{M} = \mathbf{0}$ , then

$$w = w_0 \equiv \frac{1}{2} \boldsymbol{\varepsilon} : \mathbf{C}_1 : \boldsymbol{\varepsilon} + \frac{1}{2} \boldsymbol{\varkappa} : \mathbf{C}_2 : \boldsymbol{\varkappa}.$$

# Rayleigh Principle

The modes of the shell eigen-oscillations are stationary points of the energy functional

$$\mathcal{E}[\mathbf{W}, \Psi] = \iint_{\sigma} [w_1(\boldsymbol{\varepsilon}, \boldsymbol{\varkappa}) + w_2(\Psi, \boldsymbol{\varepsilon}, \boldsymbol{\varkappa})] d\sigma, \quad (47)$$

where

$$\boldsymbol{\varepsilon} = \nabla_{\chi} \mathbf{W} + \mathbf{A} \times \Psi, \quad \boldsymbol{\varkappa} = \nabla_{\chi} \Psi,$$

on the set of functions that satisfy the kinematic boundary conditions

$$\text{on } \ell_1 : \mathbf{W} = \mathbf{0} \quad \text{and on } \ell_3 : \Psi = \mathbf{0} \quad (48)$$

and the restriction

$$\mathcal{K}(\mathbf{W}, \Psi) \equiv \frac{1}{2} \iint_{\sigma} \rho (\mathbf{W} \cdot \mathbf{W} + \gamma \Psi \cdot \Psi) d\sigma = 1. \quad (49)$$

Here the functions  $\mathbf{W}$ ,  $\Psi$  are the oscillation amplitudes for the translations and rotations, respectively.

# Rayleigh quotient

The Rayleigh variational principle is equivalent to the stationary principle for the Rayleigh quotient

$$\mathcal{R}[\mathbf{W}, \Psi] = \frac{\mathcal{E}[\mathbf{W}, \Psi]}{\mathcal{K}(\mathbf{W}, \Psi)}, \quad (50)$$

that is defined on kinematically admissible functions  $\mathbf{W}, \Psi$ .

The proof is standard and mimics one which can be found for example in the textbook<sup>7</sup>.

The Rayleigh quotient of the shell without initial stresses is

$$\mathcal{R}_0[\mathbf{W}, \Psi] = \frac{\mathcal{E}_0[\mathbf{W}, \Psi]}{\mathcal{K}(\mathbf{W}, \Psi)}, \quad \mathcal{E}_0[\mathbf{W}, \Psi] = \iint_{\sigma} w_0(\boldsymbol{\varepsilon}, \boldsymbol{\kappa}) \, d\sigma. \quad (51)$$

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<sup>7</sup>V. L. Berdichevsky, Variational Principles of Continuum Mechanics. I. Fundamentals, Springer, Heidelberg, 2009.

# Comparison of Eigen-Frequencies

Least squared eigenfrequencies are

$$\Omega_{\min}^2 = \inf \mathcal{R}[\mathbf{W}, \Psi], \quad \Omega_{0\min}^2 = \inf \mathcal{R}_0[\mathbf{W}, \Psi].$$

By the Courant minimax principle<sup>8</sup>, the Rayleigh quotient (50) allows us to estimate the values of higher eigen-frequencies. It is obvious that difference between  $\mathcal{E}$  and  $\mathcal{E}_0$  consist of two terms: difference in elastic moduli, that is the difference between  $\mathbf{C}_1$  and  $\mathbf{H}_1$ ,  $\mathbf{C}_2$  and  $\mathbf{H}_2$ , and the term  $w_2$  depending on initial stress and couple stress tensors.

**Result.** If  $w(\Psi, \boldsymbol{\varepsilon}, \boldsymbol{\kappa}) \leq w_0(\boldsymbol{\varepsilon}, \boldsymbol{\kappa})$  then

$$\Omega_k \leq \Omega_k^0, \quad k = 1, 2, \dots$$

and vice versa.

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<sup>8</sup>R. Courant, D. Hilbert, Methods of Mathematical Physics, Vol. 1, Wiley, New York, 1991.

# Uniform stretching

Let us assume that  $\mathbf{C}_1 \approx \mathbf{H}_1$ ,  $\mathbf{C}_2 \approx \mathbf{H}_2$  and uniform stretching of the shell with  $\mathbf{T} = p\mathbf{A}$ ,  $\mathbf{M} = \mathbf{0}$ ,  $p$  is the uniform tension. Here we have

$$w - w_0 = w_2.$$

We have

$$\begin{aligned}w_2(\boldsymbol{\Psi}, \boldsymbol{\varepsilon}, \boldsymbol{\kappa}) &= p \operatorname{tr} (\boldsymbol{\Psi} \times \mathbf{A} \cdot \boldsymbol{\varepsilon}) - \frac{p}{2} \operatorname{tr} (\boldsymbol{\Psi} \times \mathbf{A} \times \boldsymbol{\Psi}) \\ &= p \operatorname{tr} (\boldsymbol{\Psi} \times \nabla_{\chi} \mathbf{W}) + \frac{p}{2} \operatorname{tr} (\boldsymbol{\Psi} \times \mathbf{A} \times \boldsymbol{\Psi}) \\ &= p \operatorname{tr} (\boldsymbol{\Psi} \times \nabla_{\chi} \mathbf{W}) + \frac{p}{2} [\boldsymbol{\Psi} \cdot \boldsymbol{\Psi} + (\boldsymbol{\Psi} \cdot \mathbf{N})^2].\end{aligned}$$

Assuming  $\nabla_{\chi} \mathbf{W} = \mathbf{0}$  we obtain

$$w_2 = p/2 [\boldsymbol{\Psi} \cdot \boldsymbol{\Psi} + (\boldsymbol{\Psi} \cdot \mathbf{N})^2]$$

and the sign of  $w_2$  coincides with the sign of  $p$ . This case is similar to the dependence of eigen-frequency of a string on its tension:

stretching ( $p > 0$ ) leads to increase while compression ( $p < 0$ ) leads to decrease of eigen-frequencies in comparison with undeformable shell

# Spherical shell under hydrostatic pressure

Here  $\mathbf{f} = -q\mathbf{N}$ ,  $\mathbf{m} = \mathbf{0}$ . We introduce the Lagrangian and Eulerian spherical coordinates

$$X_1 = R \cos \Phi \cos \Theta, \quad X_2 = R \sin \Phi \cos \Theta, \quad X_3 = R \sin \Theta,$$

$$\mathbf{e}_R = (\mathbf{i}_1 \cos \Phi + \mathbf{i}_2 \sin \Phi) \cos \Theta + \mathbf{i}_3 \sin \Theta,$$

$$\mathbf{e}_\Phi = -\mathbf{i}_1 \sin \Phi + \mathbf{i}_2 \cos \Phi,$$

$$\mathbf{e}_\Theta = -(\mathbf{i}_1 \cos \Phi + \mathbf{i}_2 \sin \Phi) \sin \Theta + \mathbf{i}_3 \cos \Theta,$$

$$x_1 = r \cos \phi \cos \theta, \quad x_2 = r \sin \phi \cos \theta, \quad x_3 = r \sin \theta,$$

$$\mathbf{e}_r = (\mathbf{i}_1 \cos \phi + \mathbf{i}_2 \sin \phi) \cos \theta + \mathbf{i}_3 \sin \theta,$$

$$\mathbf{e}_\phi = -\mathbf{i}_1 \sin \phi + \mathbf{i}_2 \cos \phi,$$

$$\mathbf{e}_\theta = -(\mathbf{i}_1 \cos \phi + \mathbf{i}_2 \sin \phi) \sin \theta + \mathbf{i}_3 \cos \theta,$$

As a result we assume the deformation as

$$r = a, \quad \phi = \Phi, \quad \theta = \Theta. \quad (52)$$

$$\mathbf{Q} = \mathbf{I}. \quad (53)$$

## Spherical shell under hydrostatic pressure (2)

Using the definition

$$\nabla_{\mathcal{X}} = \frac{1}{a_0 \cos \Theta} \mathbf{e}_{\Phi} \frac{\partial}{\partial \Phi} + \frac{1}{a_0} \mathbf{e}_{\Theta} \frac{\partial}{\partial \Theta}$$

we obtain that

$$\mathbf{F} = \frac{a}{a_0} (\mathbf{e}_{\Phi} \otimes \mathbf{e}_{\phi} + \mathbf{e}_{\Theta} \otimes \mathbf{e}_{\theta}) = \lambda \mathbf{A}, \quad \lambda = \frac{a}{a_0}, \quad (54)$$

$$\mathbf{E} = (\lambda - 1) \mathbf{A}, \quad \mathbf{K} = \mathbf{0}, \quad (55)$$

and

$$\mathbf{T}_{\mathcal{X}} = (2\alpha_1 + \alpha_2 + \alpha_3)(\lambda - 1) \mathbf{A}, \quad \mathbf{M}_{\mathcal{X}} = \mathbf{0}. \quad (56)$$

Equilibrium conditions reduce to one algebraic equation  $T = q$ , where  $T = (2\alpha_1 + \alpha_2 + \alpha_3)(\lambda - 1)$ . We find that

$$\mathbf{H}_1 = \mathbf{C}_1, \quad \mathbf{H}_2 = \mathbf{C}_2. \quad (57)$$

As a result we have  $w_1 = w_0$  while

$$w_2(\boldsymbol{\psi}, \boldsymbol{\varepsilon}, \boldsymbol{\varkappa}) = T \operatorname{tr} (\boldsymbol{\psi} \times \nabla_{\mathcal{X}} \mathbf{W}) + \frac{T}{2} [\boldsymbol{\psi} \cdot \boldsymbol{\psi} + (\boldsymbol{\psi} \cdot \mathbf{N})^2].$$

# Eversion of a spherical shell

The eversion of the spherical shell is described by

$$r = a, \quad \phi = \Phi, \quad \theta = -\Theta, \quad (58)$$

$$\mathbf{Q} = \mathbf{e}_\Phi \otimes \mathbf{e}_\phi - \mathbf{e}_R \otimes \mathbf{e}_r - \mathbf{e}_\Theta \otimes \mathbf{e}_\theta. \quad (59)$$

Here

$$\mathbf{F} = \lambda(\mathbf{e}_\Phi \otimes \mathbf{e}_\phi - \mathbf{e}_\Theta \otimes \mathbf{e}_\theta), \quad \lambda = \frac{a}{a_0}, \quad (60)$$

$$\mathbf{E} = (\lambda - 1)\mathbf{A}, \quad \mathbf{K} = \frac{2}{a_0}(\mathbf{e}_\Theta \otimes \mathbf{e}_\Phi - \mathbf{e}_\Phi \otimes \mathbf{e}_\Theta), \quad (61)$$

$$\text{and } \mathbf{S}_1 = T\mathbf{A}, \quad \mathbf{S}_2 = (\beta_3 - \beta_2)\mathbf{K}. \quad (62)$$

Equilibrium conditions reduce to  $T = 0$  while

$$\mathbf{M} = \frac{2}{a_0}(\beta_3 - \beta_1)(\mathbf{e}_\phi \otimes \mathbf{e}_\theta - \mathbf{e}_\theta \otimes \mathbf{e}_\phi) \neq \mathbf{0}.$$

The eversion of a spherical shell demonstrates an example of zero initial stresses with non-zero initial couple stresses ( $\mathbf{f} = \mathbf{0} = \mathbf{m}$ ).

## Eversion of a spherical shell (2)

For such initial stressed state we obtain again that

$$\mathbf{H}_1 = \mathbf{C}_1, \quad \mathbf{H}_2 = \mathbf{C}_2$$

and

$$w_1 = w_0, \quad w_2(\boldsymbol{\psi}, \boldsymbol{\varepsilon}, \boldsymbol{\varkappa}) = \frac{1}{2} \text{tr} (\boldsymbol{\psi} \times \mathbf{M}^T \cdot \boldsymbol{\varkappa}). \quad (63)$$

Unlike to the case of the shell loaded by initial pressure for eversion there is not an external loading parameter.

### Conclusion

In both considered cases we proved that the changes in eigen-frequencies spectrum due to initial stresses are determined by initial stress fields  $\mathbf{T}$  and  $\mathbf{M}$ .

## Courant's principle: preliminary notations

To formulate Courant's minimax principle, we introduce the space  $\mathbf{H} = \{\mathbf{h} = (\mathbf{W}, \Psi)\}$  of functions  $\mathbf{h}$  satisfying (48) with respect to the energy norm defined by

$$\|\mathbf{h}\|_{\mathbf{H}}^2 = \frac{1}{2} \iint_{\sigma} (\boldsymbol{\varepsilon} : \mathbf{D}_1 : \boldsymbol{\varepsilon} + \boldsymbol{\varepsilon} : \mathbf{D}_2 : \boldsymbol{\varkappa} + \boldsymbol{\varkappa} : \mathbf{D}_3 : \boldsymbol{\varkappa}) d\sigma. \quad (64)$$

Let the eigenvalues for the problem under consideration be ordered as  $0 < \Omega_{\min} = \Omega_1 \leq \Omega_2 \leq \dots$ . To each  $\Omega_k$  there corresponds a unique eigenmode  $\mathbf{h}_k$ .

Let  $k > 1$ , and denote by  $\mathbf{H}^{(k)}$  the subspace of  $\mathbf{H}$  spanned by  $k - 1$  arbitrarily chosen elements  $\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}$  of  $\mathbf{H}$ . The space  $\mathbf{H}_{\perp}^{(k)}$  is its "orthogonal" complement in  $\mathbf{H}$ :

$$\mathbf{H}_{\perp}^{(k)} = \{\mathbf{h} \in \mathbf{H} \mid \langle \mathbf{h}, \mathbf{g}_1 \rangle_{\mathbf{L}} = \langle \mathbf{h}, \mathbf{g}_2 \rangle_{\mathbf{L}} = \dots = \langle \mathbf{h}, \mathbf{g}_{k-1} \rangle_{\mathbf{L}} = 0\},$$

$$\langle \mathbf{g}, \mathbf{h} \rangle_{\mathbf{L}} \equiv \langle (\mathbf{W}_g, \Psi_g), (\mathbf{W}_h, \Psi_h) \rangle_{\mathbf{L}} = \frac{1}{2} \iint \rho (\mathbf{W}_h \cdot \mathbf{W}_g + \gamma \Psi_h \cdot \Psi_g) d\sigma.$$

## Courant's principle: higher eigenfrequencies

$\mathbf{H}_\perp^{(k)}$  is a closed subspace of Hilbert space  $\mathbf{H}$  as the orthogonal complement to the finite dimensional space spanned by  $\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}$ . By  $\widehat{\mathbf{H}}_\perp^{(k)}$  we denote the subset of elements of  $\mathbf{H}_\perp^{(k)}$  with the constraint  $\langle \mathbf{h}, \mathbf{h} \rangle_{\mathbf{L}} = 1$ , i.e.,  $\widehat{\mathbf{H}}_\perp^{(k)} = \{ \mathbf{h} \in \mathbf{H}_\perp^{(k)} \mid \langle \mathbf{h}, \mathbf{h} \rangle_{\mathbf{L}} = 1 \}$ . Let us introduce  $d[\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}] = \inf_{\widehat{\mathbf{H}}_\perp^{(k)}} \mathcal{R}(\mathbf{h})$ .

### Courant's minimax principle

states that by taking the suprema of the quantities  $d[\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}]$  over all possible combinations  $\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}$  in  $\mathbf{H}$ , we obtain the eigenfrequencies

$$\Omega_k^2 = \sup_{\mathbf{g}_1, \dots, \mathbf{g}_{k-1}} d[\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}] = \sup_{\mathbf{g}_1, \dots, \mathbf{g}_{k-1}} \inf_{\widehat{\mathbf{H}}_\perp^{(k)}} \mathcal{R}(\mathbf{h}). \quad (65)$$

These maximum-minimum values are attained if the elements  $\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}$  coincide with the first  $k - 1$  eigenmodes.

# General remarks 1.

If

$$\mathcal{R}_1 \geq \mathcal{R}_2 \quad \text{for all } \mathbf{h} \in \mathbf{H} \quad (66)$$

then

$$\Omega_{1k} \geq \Omega_{2k} \quad (k = 1, 2, \dots). \quad (67)$$

Indeed, it follows that for any set  $\mathbf{g}_1, \mathbf{g}_2, \dots, \mathbf{g}_{k-1}$  we have

$$\inf_{\widehat{\mathbf{H}}_{\perp}^{(k)}} \mathcal{R}_1(\mathbf{h}) \geq \inf_{\widehat{\mathbf{H}}_{\perp}^{(k)}} \mathcal{R}_2(\mathbf{h}).$$

Hence

$$\Omega_{1k}^2 = \sup_{\mathbf{g}_1, \dots, \mathbf{g}_{k-1}} \inf_{\widehat{\mathbf{H}}_{\perp}^{(k)}} \mathcal{R}_1(\mathbf{h}) \geq \sup_{\mathbf{g}_1, \dots, \mathbf{g}_{k-1}} \inf_{\widehat{\mathbf{H}}_{\perp}^{(k)}} \mathcal{R}_2(\mathbf{h}) = \Omega_{2k}^2.$$

## General remarks 2.

Let  $\mathbf{H}$  be the space over which we minimize Rayleigh's quotient for a shell, and let  $\mathbf{H}_1$  be the corresponding space for the same shell subjected to a constraint of geometrical nature. Then

$$\mathbf{H}_1 \subset \mathbf{H}.$$

Here we also can demonstrate that the least eigenfrequencies satisfy  $\Omega_{1\min} \geq \Omega_{\min}$ . Using Courant's minimax principle, we can prove that

$$\Omega_{1k} \geq \Omega_k$$

for  $k = 1, 2, \dots$

# Dependence on boundary conditions

Let us consider three boundary-value problems, all with a portion  $\ell_1$  of the shell boundary clamped.

- 1 Portion  $\ell_2$  of edge free:

$$\mathbf{W}|_{\ell_1} = \mathbf{0}, \quad \Psi|_{\ell_1} = 0, \quad \boldsymbol{\eta} \cdot \boldsymbol{\Phi}_1|_{\ell_2} = 0, \quad \boldsymbol{\eta} \cdot \boldsymbol{\Phi}_2|_{\ell_2} = 0,$$

- 2 Portion  $\ell_2$  of edge simply supported:

$$\mathbf{W}|_{\ell} = \mathbf{0}, \quad \Psi|_{\ell_1} = 0, \quad \boldsymbol{\eta} \cdot \boldsymbol{\Phi}_2|_{\ell_2} = 0.$$

- 3 Entire edge  $\ell$  clamped:

$$\mathbf{W}|_{\ell} = \mathbf{0}, \quad \Psi|_{\ell} = 0.$$

By Courant's principle, one can easily establish that

$$\Omega_k^{(3)} \geq \Omega_k^{(2)} \geq \Omega_k^{(1)} \quad (k = 1, 2, \dots)$$

where  $\Omega_k^{(i)}$  is the  $k$ th eigenfrequency for the  $i$ th problem.

## Dependence on rotational inertia

If  $K_1 \geq K_2$ , then the corresponding eigenfrequencies satisfy  $\Omega_k^{(1)} \leq \Omega_k^{(2)}$  for  $k = 1, 2, \dots$

Let us analyze how the rotational inertia affects the shell eigenfrequencies. Instead of (12), we use the following form of the kinetic energy:

$$K = \frac{1}{2} \mathbf{v} \cdot \mathbf{v} + \frac{1}{2} \boldsymbol{\omega} \cdot \Theta \cdot \boldsymbol{\omega},$$

where  $\Theta$  is a positive definite second-order tensor of rotational inertia and  $\theta_1 \leq \theta_2 \leq \theta_3$ , we get

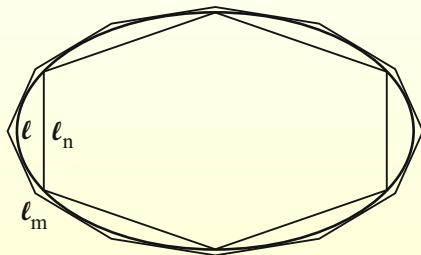
$$\theta_1 \boldsymbol{\omega} \cdot \boldsymbol{\omega} \leq \boldsymbol{\omega} \cdot \Theta \cdot \boldsymbol{\omega} \leq \theta_3 \boldsymbol{\omega} \cdot \boldsymbol{\omega}.$$

Then the eigenfrequencies for a shell where the kinetic energy includes the inertia tensor  $\Theta$  are bounded below by the eigenfrequencies calculated for shells with scalar inertia measures  $\tilde{\gamma} = \theta_1$  and  $\hat{\gamma} = \theta_3$  respectively:

$$\hat{\Omega}_k \leq \Omega_k \leq \tilde{\Omega}_k \quad (k = 1, 2, \dots).$$

# Boundary approximations

Let us consider a plate with a clamped boundary  $\ell$  and two approximated polygonal contours  $\ell_n$  and  $\ell_m$ , where  $\ell_n$  is an inscribed polygon,  $\ell_m$  is a circumscribed polygon, and  $n$  and  $m$  are the numbers of sides of  $\ell_n$  and  $\ell_m$ , respectively (Fig. 1).



**Figure:** Elliptic plate contour  $\ell$ , inscribed polygon contour  $\ell_n$  and circumscribed polygon contour  $\ell_m$  for  $n = 6$ ,  $m = 12$

# Boundary approximations

We consider three problems with the respective boundary conditions

$$\mathbf{W}|_{\ell} = \mathbf{0}, \quad \Psi|_{\ell} = \mathbf{0}, \quad \mathbf{W}|_{\ell_n} = \mathbf{0}, \quad \Psi|_{\ell_n} = \mathbf{0}, \quad \mathbf{W}|_{\ell_m} = \mathbf{0}, \quad \Psi|_{\ell_m} = \mathbf{0}.$$

Obviously, the corresponding function spaces  $\mathbf{H}$ ,  $\mathbf{H}^{(n)}$ , and  $\mathbf{H}^{(m)}$  are related as follows:

$$\mathbf{H}^{(n)} \subset \mathbf{H} \subset \mathbf{H}^{(m)}.$$

Indeed, any element  $\mathbf{h}^n \in \mathbf{H}^{(n)}$ , extended by zero, lies in  $\mathbf{H} \cap \mathbf{H}^{(m)}$ , while any element  $\mathbf{h} \in \mathbf{H}$  lies in  $\mathbf{H}^{(m)}$ . Thus we have the relations

$$\Omega_k^{(n)} \geq \Omega_k \geq \Omega_k^{(m)}, \quad \forall n, m, \quad (k = 1, 2, \dots)$$

where  $\Omega_k^{(n)}$  and  $\Omega_k^{(m)}$  are eigenfrequencies for the contours  $\ell_n$  and  $\ell_m$ , respectively. Moreover, in this case we also find that

$$\lim_{n \rightarrow \infty} \Omega_k^{(n)} = \Omega_k, \quad \lim_{m \rightarrow \infty} \Omega_k^{(m)} = \Omega_k \quad (k = 1, 2, \dots).$$

# Shells with boundary reinforcements

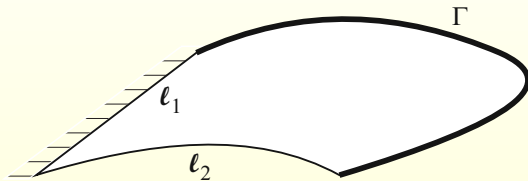


Figure: Shell with boundary reinforcement

## Shells with boundary reinforcements

For simplicity, we neglect the inertial properties of the reinforcement but do include tangential, bending, and torsional stiffness parameters. The corresponding boundary conditions are

$$\mathbf{W}|_{\ell_1} = \mathbf{0}, \quad \Psi|_{\ell_1} = \mathbf{0}, \quad \boldsymbol{\eta} \cdot \Phi_1|_{\ell_2} = \mathbf{0}, \quad \boldsymbol{\eta} \cdot \Phi_2|_{\ell_2} = \mathbf{0},$$

$$\mathbf{W}|_{\Gamma} = \mathbf{W}_{\Gamma}, \quad \Psi|_{\Gamma} = \Psi_{\Gamma},$$

and

$$\mathbf{t}' + \boldsymbol{\eta} \cdot \Phi_1|_{\Gamma} = \mathbf{0}, \quad \mathbf{m}' + \boldsymbol{\rho}'|_{\Gamma} \times \mathbf{t} + \boldsymbol{\eta} \cdot \Phi_2|_{\Gamma} = \mathbf{0},$$

where  $\mathbf{W}_{\Gamma}$  and  $\Psi_{\Gamma}$  are the translation and rotation vectors characterizing the Cosserat curve  $\Gamma$ ,  $\mathbf{t}$  and  $\mathbf{m}$  are the stress resultant and stress couple vectors, respectively, acting in the reinforcements, and a prime denotes differentiation with respect to the arc-length parameter  $s$  along  $\Gamma$ .

# Shells with boundary reinforcements

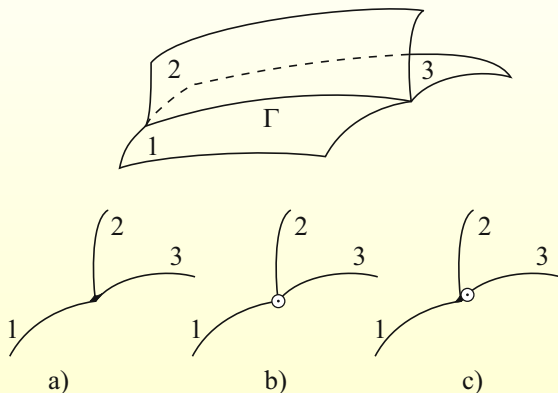
Finally, we can prove that

$$\Omega_k^{\circ} \geq \Omega_k^{\Gamma} \geq \Omega_k^f, \quad (k = 1, 2, \dots)$$

where  $\Omega_k^{\circ}$ ,  $\Omega_k^{\Gamma}$ , and  $\Omega_k^f$  are the  $k$ th eigenfrequencies of shells having the same shape but with contour  $\Gamma$  clamped, elastically reinforced, and free, respectively.

# Shells with junctions

Pietraszkiewicz and Konopińska (2011) suggested a classification of junctions for multifolded shells. Figure depicts a shell with three branches.



**Figure:** Multifolded shell and various junctions: (a) stiff junction; (b) entirely simple supported junction; (c) partially simple supported junction

## Shells with junctions

For the  $j$ th branch ( $j = 1, 2, 3$ ) emanating from the junction, we denote by  $\mathbf{W}_j$  and  $\Psi_j$  the point translations and rotations, respectively. For any type of junction, the translations are assumed to match along  $\Gamma$ :

$$\mathbf{W}_1|_{\Gamma} = \mathbf{W}_2|_{\Gamma} = \mathbf{W}_3|_{\Gamma}.$$

For the stiff junction of Fig. 3(a), the rotations are also continuous:

$$\Psi_1|_{\Gamma} = \Psi_2|_{\Gamma} = \Psi_3|_{\Gamma}.$$

For the partially simple supported junction of Fig. 3(c), we have

$$\Psi_1|_{\Gamma} = \Psi_2|_{\Gamma}.$$

For the entirely simple supported junction of Fig. 3(b), on the other hand, there are no constraints on  $\Psi_j$ . For all of these junctions, Courant's principle easily yields

$$\tilde{\Omega}_k \geq \hat{\Omega}_k \geq \bar{\Omega}_k \quad (k = 1, 2, \dots)$$

# Conclusions

- We formulated the Rayleigh variational principle for eigen-oscillations of the prestressed elastic shells.
- Using the Rayleigh quotient we shown that influence of initial stresses on the eigen-frequencies are determined by the changes of the elastic moduli tensors due to deformations and by the term depending on initial stress and couple stress tensors only.
- The latter term may play more important role in the case of flexible thin shells.
- Some specific examples are considered.

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**Thank you for your attention!**

Further Questions:

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